# **ORIGINAL PAPER**

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# Low-velocity impact response of hybrid core sandwich panels with spring and strut cores flled with resin, silicone, and foam

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# **Abstract**

Advancements in the load-bearing capacity of composite panels open doors to high-performance applications. The integration of additive manufacturing allows for the creation of intricate core designs efortlessly. Hybrid cores, combining structural elements with infll materials, play a crucial role in enhancing panel impact resistance while maintaining its low weight. This study compares sandwich panels incorporating spring and octet strut structural elements infused with diferent materials—silicon, foam, and epoxy resin—evaluating their energy absorption capabilities. Additive manufacturing is employed to produce these panels with structural elements then subsequently flled with inflls. The drop tower test is utilized to experimentally assess panel behavior under low-velocity impact. Design of experiments and statistical analysis are used to examine the infuence of core height, impact height, core geometry, and flling type on the damaged area and impactor penetration. Results showed that the strut-based structure performed better than other structures in preventing penetration, with a damaged area reduction from 501.45 to 301.58 m<sup>2</sup> compared to the spring core. The addition of foam or silicon reduced the impact damage to the front and the back sheets, with silicon inflls proving to be the most efective, reducing penetration by reducing penetration by about 60%. The depth of impact was measured, with results indicating that the truss core displayed the smallest specifc depth of penetration. A decision tree model predicted that a sandwich panel with a spring core would have a 100% chance of perforation while a flled core showed a signifcantly reduced penetration risk.

**Keywords** Sandwich panels, Low-velocity impact, Composite material, Additive manufacturing, Filling material

# **Introduction and literature review**

An ongoing effort is dedicated to designing structures that are both lightweight and high-strength, incorporating multifunctional features. Sandwich panels have appeared as a central focus in several engineering disciplines because of their exceptional properties. These complex structures, made of three layers with robust outer shells enclosing lightweight cores, result in structures that exhibit durability, impact resistance, and resilience to weather conditions.

The notable strength-to-weight ratios of sandwich panels make them versatile for applications in the construction, automotive, marine, and aerospace industries (Fischer  $2015$ ). The sandwich panel's load-bearing capabilities also contribute to high-performance set-ups, and the insulative properties of their cores support energy conservation, promoting sustainability and innovation in a world that increasingly prioritizes efficiency and environmental consciousness.

The structure of the core in sandwich panels signifcantly afects their mechanical performance, specifcally under specifc loads (Tran and Peng [2021;](#page-15-1) Ma



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et al., [2024\)](#page-15-2). While the face sheets contribute to the bending strength and rigidity, the core's primary role is to transmit the shear forces between them. Recent research has focused on optimizing energy absorption and enhancing the resistance of sandwich panels, especially in low-impact loading conditions. Advancements in 3D printing and precision cutting have allowed a new era of structural design, demonstrated by truss core sandwich panels. These panels, with core designs that are cell-based, excel in transferring the loads within outer skins which signifcantly reduces the shear defection and yields lightweight, engineered panels (Djama et al., [2020](#page-15-3)). The development of additive manufacturing and 3D printing technologies enabled the fabrication of complex core geometries, which improves load resistance and energy absorption. Core properties, such as stifness and robustness, depend on dimensions, design, thickness, and material choice.

This customization allows for the development of lightweight yet mechanically robust materials (Du Plessis et al., [2022](#page-15-4)). Manipulating the composite's composition and structure supports structural engineering principles like the truss theory, increasing the potential for tailored material designs to meet specifc application demands (Wu et al., [2023](#page-15-5); Ma et al., [2024\)](#page-15-2) investigated the ballistic impact resistance of PLA 3D-printed sandwich panels with cubic, grid, gyroid, and honeycomb infll patterns and tested them under three impact velocities: 109.95, 173, 97, and 209.48  $m/s$ . The study found that panels with cubic infll patterns exhibited the highest maximum impact load and energy absorption, performing better than the other patterns, Specifcally, the cubic and gyroid patterns showed superior impact resistance, with the cubic pattern absorbing the most energy. The research highlights the signifcant role of infll patterns in enhancing impact resistance and 3D-printed sandwich panels.

Birman and Kardomateas ([2018](#page-15-6)) classifed lattice truss cores into fve diferent types: pyramidal, corrugated, X-shaped, tetrahedral, and Kagome. O'Leary et al. ([2019](#page-15-7)) studied the mechanical behavior of these cores, with a focus on shear and compression, of Kagome truss core sandwich panels made from titanium using selective laser melting. In a similar study, Mei et al. ([2017\)](#page-15-8) examined carbon fber-reinforced polymer (CFRP) sandwich panels with a tetrahedral truss core manufactured through thermal welding in a thermoplastic matrix. The authors' main goal was to validate advanced manufacturing methods for such structures. Strut-based designs, including the kelvin cell and the octet truss structures, have shown notable mechanical properties, which enhance energy absorption at high velocities and improve impact resistance

(Hawreliak et al., [2016](#page-15-9)). Hassan et al. [\(2021\)](#page-15-10) compared the conventional honeycomb structure with the 3D-printed octet structure, proving its potential for the production of composites. The octet structure enhances the composite strength and stifness which makes it costefective for high-strength and impact applications. Liu et al. [\(2020](#page-15-11)) developed spring-cylinder, spring-rubber, and spring-aluminum foam isolation devices with damping ratios of 0.125 to 0.132. These devices perform well in resisting explosions and impact loads, achieving over 92% vibration isolation.

The adaptability of sandwich panels allows diverse material combinations that are tailored to meet specifc performance requirements (Larsson et al., [2023](#page-15-12)). This customization includes the introduction of vari-ous core fillings (Li et al., [2020](#page-15-13)), like foams and resins (Mohammadiha and Ghariblu [2016;](#page-15-14) Hassanpour Roudbeneh et al., [2019;](#page-15-15) Deng et al., [2022\)](#page-15-16), for enhancing the mechanical properties, energy absorption, and resistance to external forces. Open-cell truss core confgurations provide easy access for integrating lightweight materials to further enhance the panel performance.

Researchers have studied flling metallic sandwich panels with polymer or metallic foams for cost-efectiveness. Foam integrated into the core signifcantly enhances the compression strength, shear strength, energy absorption, perforation resistance, and blast resistance (Yan et al., [2013](#page-15-17); Bin et al., [2015;](#page-15-18) Karttunen et al., [2017](#page-15-19)) observed improved fatigue strength in foam-flled sand-wich panels. (Najafi and Eslami-Farsani [2021\)](#page-15-20) enhanced the structural and durability properties of PU foambased sandwich panels without increasing the weight. The design incorporated through-thickness stiffening, core hybridization, and improved core-skin interfaces. This was done through the use of agglomerated cork/PU foam and lattice structures. The novel panels portrayed a 506% increase in the maximum fexural load and an 816% increase in the initial fexural stifness as compared to traditional panels. Moreover, the novel panels absorbed 21% more impact energy and demonstrated better damage tolerance and energy absorption.

Mei et al. ([2022](#page-15-21)) conducted three-point bending tests on polyurethane foam-flled CFRP X-core sandwich panels and reported a signifcant increase in the initial failure and peak load. Wang et al. ([2021](#page-15-22)) enhanced the compression strength and bending resistance in lattice truss sandwich panels using polyurethane and concrete foam fllings. Zhang et al. ([2023](#page-16-0)) found that introducing polyurethane foam in Kirigami corrugated core panels increased the energy absorption by 88.5% for blast mitigation.

Deng et al. ([2022](#page-15-16)) assessed the ballistic properties of an S-shaped folded sandwich confguration with microbead modifed epoxy resin in the core. Yuan et al. ([2018](#page-16-1)) experimented with silicone resin and carbon powder as fllers for enhancing the mechanical properties of the truss. Introducing thermal insulation and ablative materials extended the failure time and reduced the ablation damage. Warren et al. ([2021](#page-15-23)) studied honeycomb core sandwich panels flled with a shear-thickening fuid under hypervelocity impact, signifcantly decreasing the damage levels and preventing complete perforation. Noor Azammi et al. [\(2018\)](#page-15-24) blended kenaf fber with natural rubber and thermoplastic polyurethane to enhance mechanical properties, such as tensile strength and fexural strength. The reinforced specimen outperformed normal composites.

To further explore the enhancement of impact resistance in sandwich panels, Maher et al. [\(2022\)](#page-15-25) investigated the high-velocity impact behavior of corrugated core composite sandwich panels with composite face sheets reinforced by pseudo-elastic NiTi shape memory alloy (SMA) wires. The study aimed to improve the impact resistance by incorporating SMA wires into the composite face sheets. Results demonstrated that the addition of SMA wires signifcantly reduced the projectile's residual velocity and that pre-strained wires markedly enhanced energy absorption. The location of SMA wires at the impact site played a crucial role, leading to better energy absorption compared to wires placed away from the impact site. The inclusion of SMA wires enhanced structural integrity and impact resistance, making them suitable for applications requiring high energy absorption and damage tolerance.

This research proposes an innovative composite structure using 3D additive manufacturing to integrate a structural core with a polymeric flling, aiming to enhance the performance and redefne composite design. Physical experiments will compare the impact of spring and strut core structures with polyurethane foam, epoxy resin, and silicone rubber fllings on the performance of composites. A drop tower test will evaluate the panel's behavior under low-velocity impact, assessing the impact absorption capabilities of various sandwich panels with diferent core designs and fllings. Regression analysis is applied to model and predict panel behavior, providing insights for material design in critical industries like aerospace and automotive. Classifcation modeling with decision trees is used to predict the penetration probability, highlighting safety implications for practical applications. Traditional materials often struggle to meet the combined requirements of strength, weight, and functionality.

By taking advantage of advancements in additive manufacturing and new materials, this study explores novel design possibilities, pursuing performance enhancements through customized core geometries and fllings.

## **Materials and methods**

Figure [1](#page-2-0) shows the sandwich panel dimensions, with a fixed length and width of 80 mm. The core height (h) varied, while the upper and lower face sheet thickness were constant at 1 mm. A cellular core volume fraction of 15% was maintained for all core designs. The study explored two core structures: octet strut and spring, as shown in Fig. [2](#page-3-0). The octet truss and spring core designs were selected for this research due to their unique structural properties, important for enhancing the performance of sandwich panels. The octet truss core is known for its exceptional strength and stiffness and can efficiently distribute the loads and reduce shear deformation, making it ideal for high-strength and impact-resistant applications. Its compatibility with additive manufacturing processes allows for the accurate fabrication of complex shapes. The spring core, on the other hand, is chosen for its ability to absorb and redistribute impact energy, due to its high in-plane elastic modulus. This core design spreads the shear forces and compression across neighboring cells which results in efective energy dissipation.

To study the cellular core's efect on energy absorption in the sandwich panels, three variables were examined: the core height, core flling type, and the impact height. A full factorial design in Table [1](#page-3-1) established the experimental runs, including two core heights (10 mm and 20 mm), three reinforcement fllings (foam, silicon, and epoxy



<span id="page-2-0"></span>**Fig. 1** Sandwich panel dimensions



<span id="page-3-0"></span>**Fig. 2** Schematic and 3D printed sandwich panels for an **a** octet truss core and **b** spring core

<span id="page-3-1"></span>**Table 1** Factorial design for experimental study

<b>Factor</b>	Levels	Values		
Core geometry	R	None, spring, strut		
Core height (mm)		10, 20		
Impact height (cm)		15.25		
Filling type		Foam, silicon, resin, none		

resin), and two impact energies achieved by changing the drop height (15 mm and 25 mm).

Additionally, a control sample without a core, assuming a rectangular confguration, served as a scale for all variable conditions. The study investigated a total of 48 unique combinations of these variables.

Architected sandwich panels were 3D printed using a Flash Forge printer (Fig. [3\)](#page-3-2), which uses fused deposition modeling (FDM) technology for creating polylactic acid (PLA) polymer samples. This process is aligned with ASTM standard F2792 (MIT [2012](#page-15-26)). FDM, involves the extrusion of heated polymeric flaments in the *x*- and *y*-coordinates, while the build table lowers the object layer by layer in the z-direction. The orientation and toolpath of each layer, as prescribed by ASTM F2792 (Baich et al., [2015](#page-15-27)), afect the mechanical properties of the printed structures.

PLA was chosen due to its simplicity and accessibility which aligns with the study's focus on topological attributes and diverse fillings. The filament diameter used was 1.75 mm from Filatech UAE. Three fillings, epoxy resin, foam, and silicon, were strategically selected for their varied mechanical properties and impact behaviors.

Polyurethane foam from Hi Stick® expanded by 50% during curing and was contained within the panel using tape (Fig. [4b](#page-4-0)). EL160 high-temperature epoxy laminating resin from Easy Composites® was carefully mixed at a ratio of 1:0.35 (resin: hardener) and injected into the cellular cores using a syringe, taped from three sides (Fig. [4](#page-4-0)a), and required 24 h to cure. Silicon from Sikasil<sup>®</sup> was chosen for its one-component nature and resistance to cracking applied using a sealant gun and to cure through atmospheric moisture. Table [2](#page-4-1) presents the

<span id="page-3-2"></span>

**Fig. 3** 3D printing of **a** spring core and **b** strut core sandwich panel



**Fig. 4** Core flling process **a** resin and **b** polyurethane foam

## <span id="page-4-1"></span><span id="page-4-0"></span>**Table 2** Material properties





**Fig. 5** Diferent flling types in the octet truss core sandwich panels

<span id="page-4-2"></span>mechanical properties of these materials. Figure [5](#page-4-2) shows the three flling types in the octet truss sandwich panel.

This study analyzed mass changes in various samples, considering core topology and flling type. Spring and strut cores had the least mass increase with resin, foam, and silicon compared to the hollow panel. The panel with no core showed the highest mass increase, as expected with an initial volume fraction of 0%. Foam

had the least impact on the mass increase, while resin had the most substantial increase due to its higher density. These findings reveal the complex relationship between core topology, flling type, and mass changes, offering insights into sandwich panel materials and design considerations.

## **Experimental setup**

Figure [6](#page-5-0)a shows the drop test tower setup for low-velocity impact experiments. It uses a 0.18 kg steel impactor connected to a 15-kg plate, dropped from 15 and 25 cm heights. Raising the drop height increases the impact energy. Impact velocities just before contact were 1.7 m/s and 2.2 m/s for the 15 cm and 25 cm heights, respectively. The impactor drops on a simply supported beam. Two responses evaluated were damaged area and impact indentation depth on both the front and back sides of the panel. To ensure reliability and consistency, two identical samples were made for each of the 48 confgurations, totaling 96 tests. Results were averaged, with a third test conducted in cases of deviations exceeding 5% for accuracy.

## **Results and discussion**

The damage area on the face sheets was measured using Adobe Photoshop Lasso's tool which traces the damage outline, records the pixel dimensions using the measurement log, and converts the measurements to real-world for precise quantifcation.

During low-velocity impact tests, some samples fully penetrated while others showed partial damage without perforation. For non-penetrating specimens, the depth of impact was precisely measured using a vernier depth gauge, indicating the panel's ability to withstand penetration and endure deformation. A smaller depth indicates higher resistance to penetration and greater capacity for energy absorption and deformation. A larger impact area suggests more extensive damage and can be correlated with the transferred energy during impact. Smaller impact areas signify efective energy dissipation and force absorption. These measurements of impact area and depth provide valuable insights into damage patterns, energy absorption capabilities, and overall mechanical behavior of sandwich panels under low-velocity impact.

Surface response analysis was used to study the results of the damaged area and depth of impact using Minitab software. The main effects plot shows the average response for each level of a factor, irrespective of the levels of other factors. Each point on the plot represents the average outcome for a particular level of a factor, averaged across all other factors included in the analysis. Figure [7](#page-6-0) depicts the main efect plots revealing the correlation between the design variables and their impact on the average damaged area on the back face sheets of the sandwich panels. The plot for the core height shows the mean damaged area for diferent core heights, averaged over all impact heights, core geometries, and flling types. Similarly, the plot for the impact height shows the mean damaged area for diferent impact heights, core geometries, and filling types. This approach provides a clear understanding of how each factor independently influences the mean damaged area. The same trend can be seen for both the front and back face sheets of the sandwich panels. As illustrated in Fig. [7,](#page-6-0) there exists a linear inverse relationship between core height and damaged area. As the core height increases, material stifness also increases, resulting in a reduction in the damaged area at the back of the panel compared to a panel with a shorter core height. This is also depicted in Fig.  $8$  showing the damaged area on the front and back face sheet of a truss core sandwich panel with foam core flling. It is



<span id="page-5-0"></span>**Fig. 6 a** Experimental setup for low-velocity impact tests. **b** 2D schematic illustrating sandwich panel and impactor height



Main Effects Plot for Damaged Area on Backside

<span id="page-6-0"></span>**Front Facesheet Back Facesheet** 10 mm core height 20 mm core height

<span id="page-6-1"></span>**Fig. 8** Truss core sandwich panel with foam flling at impact height 15 cm

evident that as the core height increases, irrespective of the impact height, the damaged area decreases.

An interaction plot visualizes the interaction efects between two or more factors on a response variable. These plots show how the mean response (damaged area) changes with the levels of one factor while considering diferent levels of another factor. Each line in the interaction plot represents a diferent level of one factor, showing how the response variable changes with the levels of another factor. Analysis of the interaction plots in Fig. [9](#page-7-0) shows a signifcant interaction between core geometry and flling material. Panels with no core flling have the highest damage, while those with a strut core show the lowest face sheet damage, regardless of flling type. Foam and silicon are the most robust fllings, consistent with the back face sheet damage.

Changing the impactor's drop height to vary the impact velocity causes a slight increase in the mean



Interaction Plot for Damaged Area on Backside<br>Fitted Means

<span id="page-7-1"></span><span id="page-7-0"></span>**Fig. 9** Interaction plot of damaged area (back) with core height, impact height, and core geometry



**Fig. 10** Truss core sandwich panel (20 mm core height) with no flling

damaged area, from 805.77 to 897.88 mm<sup>2</sup>, as shown in Fig. [7.](#page-6-0) This increase is due to the greater impact force at higher velocities. However, it is not the most signifcant factor, as indicated by the main efect plot. Higher impact energy would ensure penetration but not necessarily lead to a further extension of the damaged area, as will be demonstrated later in the study.

Figure [10](#page-7-1) shows the damaged area of the truss core sandwich panel with no flling. It can be seen that the damage area increases as the impact height increases. A similar trend is observed in all samples regardless of core topology or flling material.

Regarding core geometry, it is evident that when a sandwich panel features an entirely hollow core, it cannot withstand damage as well as other panels with cellular cores. However, it is used as a control specimen for the purpose of comparison. In the case of the strut and spring topologies, the core's volume fraction remained the same. However, the spring core resulted in a higher damaged area, decreasing from  $501.45$  to  $301.58$  mm<sup>2</sup> compared to the strut.

When comparing the topology of the strut and spring core, results showed the spring has the ability to efectively withstand impact forces by absorbing energy, primarily due to its higher in-plane elastic modulus. Moreover, the spring topology exhibits higher fexural stiffness than that of the strut. Therefore, in terms of the damaged area, the spring core redistributes both shear forces and out-of-plane compression forces across neighboring cells, which results in a larger damaged area.

As a result, the impact forces are transmitted not just to the immediate vicinity of the point of impact but also to neighboring cells within the core. This widespread load distribution causes neighboring cells to deform, resulting in a more extensive area of damage on the face sheets. In contrast, the strut core, with its higher fexural strength, confnes the damage to a localized region, resulting in a smaller afected area that is not spread across the panel. Figure [11](#page-8-0)a–c shows the damaged area of the void core, spring core, and strut core sandwich panels with no flling, respectively. It can be seen that the lowest damage was observed in the front and back face sheets of the



<span id="page-8-0"></span>**Fig. 11** Sandwich panels with a 20-mm core at an impact height of 15 cm. Note: all sandwich panel samples are manufactured from the same spool

truss core while the void core has the highest damage since the specimen broke in half.

While examining the efect of diferent fllings, it is observed that the absence of fllings results in the great-est damage as seen in Fig. [7](#page-6-0). This is primarily due to the minimal material stiffness within the cellular core. The inclusion of fllings serves the purpose of introducing an additional layer of damping material to the cellular core, contributing to its overall energy absorption capabilities.

The foam-filled samples demonstrate superior performance compared to sandwich panels flled with silicon and resin. Specifcally, the damaged area at the back side in the foam-flled sandwich panel is signifcantly reduced compared to that in the resin-flled panel, with values of 995.5  $\text{mm}^2$  and 201.34  $\text{mm}^2$ , respectively. This improvement can be attributed to the reinforcement of specimen stiffness achieved through foam filling. The crushable nature of polyurethane foam plays a crucial role in mitigating damage and dissipating the energy applied in crushing the foam inside rather than the back sheet of the panel. On the other hand, silicon outperforms epoxy resin as a core fller, even though the resin

can withstand higher in-plane forces due to its superior strength, but the superiority of silicon is attributed to its greater ductility exhibiting a higher toughness than that of the resin.

Figure [12a](#page-9-0)–d shows the damaged area of the truss core sandwich panel with no flling, resin flling, foam flling, and silicon flling, respectively. It is evident that the foam and silicon flling have the lowest damage on both face sheets while unflled panels have the highest damage.

To assess impact depth more efectively, variations in mass were considered, especially due to density differences in the fllings. Samples with resin and silicon showed a signifcant mass increase, particularly those without a core. This analysis determines the optimal sandwich panel confguration by evaluating the penetration depth relative to the weight of the samples. The optimal sample is identifed as the one with the least penetration per gram of material, indicating higher impact resistance and efficiency. To normalize the depth of impact, a specifc depth was used which indicates the depth of impact per unit mass (Eq. [1\)](#page-10-0).



<span id="page-9-0"></span>**Fig. 12** Sandwich panels at impact height 15 cm



<span id="page-10-1"></span>Specific depth of impact = 
$$
\frac{\text{Depth of impact (mm)}}{\text{Mass after filling (g)}}
$$
(1)

Figure [13](#page-10-1) shows the infuence of the core height on the specifc depth of impact, decreasing from 0.583 to 0.42 mm/g as the core height increased from 10 to 20 mm. A similar trend is seen in Fig. [15](#page-12-0) for the change of specifc depth with the change of impact height, where higher energy results in deeper penetration and a larger damaged area.

Analyzing Fig. [13](#page-10-1) shows that the truss core displayed the smallest specific depth of penetration. The truss core's superior energy absorption capacity and stiffness differentiated it from the spring core, which distributed impact depth along the sandwich panel's height. Figure [13](#page-10-1) shows that resin and silicon fillings provided the lowest depth of impact followed by foam core and then the empty core.

Comparing Figs. [13](#page-10-1) and [7](#page-6-0), it can be analyzed that the mean damage area with resin flling was high when compared to the case where the mean specifc depth was notably low. This change comes from cases where lowvelocity impact did not lead to perforation of the sandwich panels. In such cases, the impact depth was much lower than that observed in samples flled with either silicon or foam.

Figure [14](#page-11-0) shows the interaction plot for specifc depth involving the core geometry and flling type, showing their signifcant interaction. Pareto analysis in Fig. [15](#page-12-0) proves this interaction, with silicon and resin fllings <span id="page-10-0"></span>not making a diference for strut elements, whereas foam improves the structure for strut elements but not for spring elements. Figure [15a](#page-12-0) indicates that the core geometry is the most infuential factor for the damaged area in the front face sheets, with increased efects from second-order interactions. Figure [15b](#page-12-0) shows core height's second-order interactions for the damaged area in the back face sheets. For specifc depth, the flling type is the primary factor, followed by its second-order interaction with core geometry.

#### **Regression analysis**

Statistical analysis was conducted evaluating the impact of core topology, core flling, and core height on the damaged area of the front and back face sheets, and the specifc depth of impact. Regression analysis plays an important role in understanding the impact on sandwich panels by quantifying the efects of variables like core geometry, flling type, and impact height on panel damage.

Response surface methodology was used for building a regression model, excluding terms with high *P* values in the analysis of variance, resulting in linear terms and their interactions. The model achieved a good fit with  $R^2$ values of 88.46% for the damaged area front, 87.45% for the damaged area back, and 93.74% for specifc depth. Regression equations (Tables  $3, 4$  $3, 4$ , and  $5$ ) describe how core height and impact height infuence damage area and depth of impact, presented in a linear format as follows:



# Interaction Plot for Specific Depth of Impact **Fitted Means**

<span id="page-11-0"></span>**Fig. 14** Interaction plot of specifc depth with core height, impact height, and core geometry

$$
R = a + b(H_C) + c(H_I) + d(H_C)(H_I)
$$
\n(2)

where  $H_C$  and  $H_I$  are the core height (mm) and impact height (cm), respectively. a is the constant, *b* is the linear coefficient of core height, c is the linear coefficient of impact height, and d is the two-way interaction coefficient between core height and impact height. The values of these terms are provided in Tables [3,](#page-13-0) [4,](#page-13-1) and [5](#page-13-2) for each of the response variables.

#### **Classifcation modeling**

Decision tree analysis was used to model whether penetration will occur for that panel or not. Classifcation tree model (CART®) in Minitab statistical software was used for that purpose. The Classification tree model (CART) serves a pivotal role in predicting penetration events in sandwich panels, functioning as a binary classifcation tool to discern the conditions leading to material failure or impact resistance. This model's utility extends signifcantly into safety-critical applications, particularly in sectors like the aerospace industry. Understanding the

probability of penetration is key to making informed design choices that bolster aircraft safety. Manufacturers can leverage this model to identify optimal combinations of core materials and fllers, efectively minimizing penetration risks and thereby upholding higher safety standards. This practical application of the CART model demonstrates its importance in enhancing material safety and reliability in critical environments.

Results reveal the most crucial variables among the input parameters within the tree. Results of the model are given in Fig. [16](#page-14-0), indicating which conditions would result in penetration. It shows that a sandwich panel with a spring core will have a 100% chance of perforation while a strut core and a void core will have a 52.4% chance of perforation. However, if these cores are flled with either resin, silicon, or foam, they will have a 41.2% chance of penetration while it will be 100% if no fllers are used. Tables [6](#page-14-1) and [7](#page-15-28) present a detailed overview of the confusion matrix and its associated statistics, providing valuable insights into the model's classifcation performance. Notably, the highest level of accuracy was achieved for the "no penetration" event, with a perfect accuracy rate



<span id="page-12-0"></span>**Fig. 15** Pareto charts for **a** damaged area on front face sheets, **b** damaged area on back face sheets, and **c** impact depth

<span id="page-13-0"></span>**Table 3** Regression analysis equations for damage area on front face sheets

Core geometry	<b>Filling type</b>	a	b	C	d
None	Foam	2278	124	$-24.9$	1.31
Spring	Foam	283	3	$-4.6$	1.31
Strut	Foam	20	6	$-12.9$	1.31
None	None	6492	120	$-3.2$	1.31
Spring	None	$-285$	$\Omega$	17.1	1.31
Strut	None	$-263$	10	8.8	1.31
None	Resin	6445	268	13.8	1.31
Spring	Resin	2225	148	6.4	1.31
Strut	Resin	1968	138	$-1.9$	1.31
None	Silicon	1803	105	$-23.1$	1.31
Spring	Silicon	$-286$	15	$-2.8$	1.31
Strut	Silicon	17	24	$-11.1$	1.31

<span id="page-13-1"></span>**Table 4** Regression analysis equations for damage area on back face sheets

<b>Filling type</b>	a	b	c	d
Foam	1553	85	$-12.9$	0.09
Foam	461	17	$-11.9$	0.09
Foam	$-350$	7	18.4	0.09
None	6054	$-84$	10.1	0.09
None	10	18	11.1	0.09
None	$-407$	8	41.4	0.09
Resin	5657	$-243$	23.7	0.09
Resin	2327	$-142$	24.8	0.09
Resin	1554	$-151$	55.1	0.09
Silicon	2007	- 99	$-27.4$	0.09
Silicon	956	$\mathfrak{D}$	$-26.4$	0.09
Silicon	448	$-7$	3.9	0.09

<span id="page-13-2"></span>**Table 5** Regression analysis equations for specifc depth of impact



of 100%. Conversely, the event of "penetration" exhibited the lowest accuracy during the training phase, reaching 65%.

However, when considering the overall performance of the model, particularly its ability to accurately classify the "penetration" event, it is found that it performs commendably. During the training phase, the model achieved an accuracy rate of 76.7% in classifying penetration events, and this performance was maintained at 72.2% accuracy during the testing phase. These results are promising, especially when considering the size of the sample dataset, and they underscore the model's potential for robust classifcation in scenarios involving penetration events.

## **Conclusion**

In this research, a comprehensive experimental analysis was conducted to assess the energy absorption capabilities of various sandwich panels under low-velocity impact conditions. These panels featured distinct core topologies and flling materials, incorporating two cellular core designs- namely, spring and octet truss. The panels were filled with silicon, foam, or epoxy resin. A total of 48 individual samples were tested, each subjected to various combinations of independent variables. The experiments, conducted through drop tower tests, examined the impact of core geometry, core height, flling type, and impact height on the performance of the sandwich panels. Fused deposition modeling (FDM) technology was utilized to print PLA panels. Three different fillings (i.e., polyurethane foam, epoxy resin, silicon) were selected as inflls due to their unique mechanical properties. Polyurethane foam exhibits high thermal stability, epoxy resin offers high performance, and silicon possesses high toughness.

Results indicated that core height, flling type, and core geometry signifcantly infuenced the damaged area and depth of impact. Core height exhibited a linear inverse relationship with the damaged area, attributed to increased core stifness with higher core heights. Core geometry played a role in the damaged area, with the spring core recording the highest damage for panels without fllings. In terms of depth of impact, core height had the most pronounced effect, followed by impact height. The spring core results in a larger damaged area compared to the strut core since the spring core redistributes forces, leading to a more extensive area of damage. Sandwich panels flled with polyurethane foam outperformed those flled with resin and came very close to silicon in terms of reducing the damaged area at the back face sheet. However, they failed to reduce the depth of penetration compared to the resin and silicon-flled structures.

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<span id="page-14-0"></span>**Fig. 16** Decision tree for penetration of sandwich panels

<span id="page-14-1"></span>



<span id="page-15-28"></span>**Table 7** Statistics for the confusion matrix

<b>Statistics</b>	Training (%)	Test $(\% )$	
True positive rate (sensitivity or power)	65	714	
False positive rate (type I error)	$\left( \right)$	25	
False negative rate (type II error)	35	286	
True negative rate (specificity)	100	75	

Regression analysis was used to model the damaged area and depth of penetration in terms of the design factors. The models were able to predict the behavior with a high level of accuracy, achieving the largest  $R^2$  of 93.74%. Decision trees were also utilized to predict the probability of penetration occurrence based on the structural element shape and flling type yielding a confusion matrix, with an accuracy of 72.2% during the testing phase. Designers can employ the developed model to assess penetration outcomes for a particular application.

#### **Abbreviations**

CFRP Carbon-fbre reinforced polymer

FDM Fused deposition modeling

PLA Polylactic acid

#### **Authors' contributions**

AC worked on methodology, investigation, experimentation, data Analysis, and writing the frst draft. NMH worked on conceptualization, methodology, formal analysis, supervision, and editing and revision of fnal version. ZB worked on conceptualization, methodology, investigation, supervision, and editing and revision of the fnal version. MI worked on methodology and experimentation. All authors read and approved the fnal manuscript.

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#### **Availability of data and materials**

The datasets used and/or analyzed during the current study are available from the corresponding author on reasonable request.

## **Declarations**

#### **Competing interests**

The authors declare that they have no competing interests.

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